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Control Optimization for Uncertain Systems via the Koopman Operator

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 The need to make decisions under uncertainty arises often in engineering and scientific applications



 The need to make decisions under uncertainty arises often in engineering and scientific applications

| Airdrop Package Delivery Using Ballistic Parachute |
|--|
| Nominal Path |
| Predicted Dispersion |
| Q: Where should the package be dropped? |



 The need to make decisions under uncertainty arises often in engineering and scientific applications





Initial density over uncertain states and parameters

 (\boldsymbol{x})

State transformation under control selection u

 $S_u(x)$

 $P_{S}f(\boldsymbol{x})$

"Pushed-forward" density and objective function

Choose *u* that maximizes $\mathbb{E}\left[g(X) | X \sim P_S f\right] = \int_{S(\Omega)} P_S f(x) g(x) dx$



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Choose *u* that maximizes
$$\mathbb{E}\left[g(X) | X \sim P_S f\right] = \int_{S(\Omega)} P_S f(x) g(x) dx$$

Okay...but how do we compute $P_s f(x)$ for nonlinear/non-Gaussian systems?



Forward Density Propagation for Uncertain Systems

Frobenius-Perron (FP) Operator

Monte Carlo Simulation







Polynomial Chaos

$$Y = \sum_{j=0}^{p} y_{j} \psi_{j}(\Xi) = \eta(x) \qquad X_{P} = \sum_{j=0}^{p} x_{j} \psi_{j}(\Xi)$$



The Koopman Operator



Koopman Operator

 $S: \mathbb{R}^n \to \mathbb{R}^n$

State Map

- Properties of Koopman operator of a system reveals properties of the underlying system
- Recent advancement in the literature for approximating via data-driven methods
 - Extended Dynamic Mode Decomposition (Williams et al. 2014, Korda and Mezic 2018)





$$\mathbb{E}\left[\mathcal{K}g\left(X\right)|X \sim f\right] = \int_{\Omega} f\left(\boldsymbol{x}\right) \mathcal{K}g\left(\boldsymbol{x}\right) d\boldsymbol{x}$$

$$\mathbb{E}\left[g\left(X\right)|X \sim P_{S}f\right] = \int_{S(\Omega)} P_{S}f\left(\boldsymbol{x}\right)g\left(\boldsymbol{x}\right)d\boldsymbol{x}$$

Push-Forward (FP) Expectation



Halder and Bhattacharya, 2011

- Improved numerical stability
- Simpler evaluation
 - Domain of integration is initial domain Ω vs its maye $S(M^2)$
 - Provides well-defined structure of data, leading to simpler solution approaches (e.g., quadrature integration)

VS.



Meyers et al., ACC 2019.



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Simpler evaluation

Improved numerical stability

- Domain of integration is initial domain Ω vs its maye $S(M^2)$
- Provides well-defined structure of data, leading to simpler solution approaches (e.g., quadrature integration)
- Computing expectation of multiple observables with varying supports in space-time

VS.

• Pull-back each to a common domain domain \rightarrow Single, vector-valued expectation calculation

Meyers et al., ACC 2019.





- Error bounds / tolerancing via quadrature integration
- Downside: Koopman expectation assumes no process noise.
 - Application limited to systems which have only parametric uncertainty

$$\mathbb{E}\left[\mathcal{K}g\left(X\right)|X \sim f\right] = \int_{\Omega} f\left(\boldsymbol{x}\right) \mathcal{K}g\left(\boldsymbol{x}\right) d\boldsymbol{x}$$

Pull-Back (Koopman) Expectation

Generalized Polynomial Chaos (gPC)

- In general, they are not the same
- However, there is an equivalence between the Koopman expectation and non-intrusive gPC when computing the mean value of an observable function g(x)

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VS.

Koopman Expectation

$$\int_{\Omega} \mathcal{K}g(\boldsymbol{x})f(\boldsymbol{x})d\boldsymbol{x} = \int_{\Omega} \eta\left(f_p(\boldsymbol{\xi})\right)\psi_0(\boldsymbol{\xi})p(\boldsymbol{\xi})d\boldsymbol{\xi}$$
$$g\left(S\left(f_p(\boldsymbol{\xi})\right)\right) = \mathcal{K}g(\boldsymbol{x})$$

gPC computation of mean of transformed RV



$$\mathbb{E}\left[\mathcal{K}g\left(X\right)|X \sim f\right] = \int_{\Omega} f\left(\boldsymbol{x}\right) \mathcal{K}g\left(\boldsymbol{x}\right) d\boldsymbol{x}$$

Pull-Back (Koopman) Expectation

Generalized Polynomial Chaos (gPC)

- In general, they are not the same
- However, there is an equivalence between the Koopman expectation and non-intrusive gPC when computing the mean value of an observable function g(x)
- Koopman advantage: When computing higher-order moments, Koopman method (redefining observable) requires a lot less integrals than gPC

VS.

 gPC advantage: You can sample from transformed distribution (Koopman expectation only provides expected values)



Using the Koopman Expectation for Probabilistic Optimization

- In practice, we do not compute Koopman operator \mathcal{K}_S
- Instead, we compute *action* of the Koopman operator on observable functions of interest at discrete points in state space $\mathcal{K}_S g(x_i)$
 - Then integrals can be approximated via quadrature

$$\int_{\Omega} \mathcal{K}_{S}g(x)f_{0}(x)dx \approx \sum_{i=1}^{N} \mathcal{K}_{S}g(x_{i})f_{0}(x_{i})w_{i}$$

Note: We can also use other methods such as Monte Carlo integration to compute this as well.

 $\mathcal{K}_{S}g(x_{i})$: From each discrete sample x_{i} , forward simulate and compute observable function

 $f_0(x_i)$: Initial uncertainty PDF evaluated at sample x_i

w_i: Quadrature weight

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Probabilistic Optimization via the Koopman Expectation

• We wish to solve the following optimization problem:

$$\boldsymbol{u}^* = \underset{\boldsymbol{u}\in\mathcal{U}}{\arg\min} \int_{\Omega} \mathcal{K}_{S}g(\boldsymbol{x},\boldsymbol{u})f_0(\boldsymbol{x})d\boldsymbol{x}$$

Minimize expected value of cost

subject to:

 $\int \mathcal{K}_{S} \boldsymbol{c}(\boldsymbol{x}, \boldsymbol{u}) f_{0}(\boldsymbol{x}) d\boldsymbol{x} < \boldsymbol{r}$

Satisfy chance constraints



Meyers et al., ACC 2019.

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subject to:

$$\int_{\Omega} \mathcal{K}_{S} \boldsymbol{c}(\boldsymbol{x}, \boldsymbol{u}) f_{0}(\boldsymbol{x}) d\boldsymbol{x} < \boldsymbol{r}$$

Key point: Because cost and constraint functions pulled back to initial time via Koopman operator, $f_0(x)$ is never explicitly propagated forward in time.



Example 1: Bouncing Ball

Bouncing ball in 2D with uncertain coefficient of restitution. Compute expected cost value (no optimization).

System:

$$\ddot{\boldsymbol{x}} = \begin{bmatrix} \ddot{x} \\ \ddot{z} \end{bmatrix} = \begin{bmatrix} 0 \\ -g \end{bmatrix}, \quad x_0 = 2 \text{ m}, \dot{x}_0 = 2 \text{ m/s}, z_0 = 50 \text{ m}, \dot{z}_0 = 0 \text{ m/s}$$
$$\dot{z}^+ = -\alpha \dot{z}^- \quad \text{when } z = 0$$

Uncertainty:

 $\alpha \sim \mathcal{N}(0.9, 0.02)$ truncated at 0.84 and 1

Cost:

$$g\left(\boldsymbol{x}\right) = \left(z - z^*\right)^2$$



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Gerlach et al., 2020, https://arxiv.org/pdf/2008.08737.pdf

Example 1: Bouncing Ball

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| | Analytical | Monte Carlo | Koopman |
|------------------------------|------------|-------------|---------|
| No. of Simulations | - | 100,000 | 15 |
| Exp. Value (m ²) | 36.008 | 35.782 | 36.008 |
| Computation Time (s) | - | 2.060 | 0.0012 |

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Gerlach et al., 2020, https://arxiv.org/pdf/2008.08737.pdf



Example 1: Bouncing Ball

Particular observable functions defined to "extract" raw moments (which can then be converted to central moments)

$$g_1(x) = x$$
mean $g_2(x) = x^2$ 2^{nd} raw moment $\int_{\Omega} \mathcal{K}_S g_i(x, u) f_0(x) dx$ $g_3(x) = x^3$ 3^{rd} raw moment \vdots

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| Central Moment | Monte Carlo | Koopman |
|----------------|------------------------|-----------------------------------|
| 2 | $9.030 \mathrm{e}{-2}$ | $9.007e-2 \pm 3.878e-5$ |
| 3 | $3.878e{-1}$ | $3.924e{-}1 \pm 1.776e{-}3$ |
| 4 | 3.214 | $3.428 \pm 1.536 \mathrm{e}{-3}$ |
| 5 | 38.116 | $44.536 \pm 3.733 \mathrm{e}{-3}$ |
| | | |

10M simulations, 264 sec 225 simulations, 3.4 ms

Koopman-based method produces solution/ with same accuracy but runs 77,000x faster.

Gerlach et al., 2020, https://arxiv.org/pdf/2008.08737.pdf/

Example 2: Airdrop Mission Planning

High-Altitude Low-Opening (HALO) Airdrop

Leonard et al., J. Guidance, Control, and Dynamics, 2020.

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Cost Function g(x, y)

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Example 2: Airdrop Mission Planning

High-Altitude Low-Opening (HALO) Airdrop

Leonard et al., J. Guidance, Control, and Dynamics, 2020.

- **Deterministic planner** does not account for uncertainty, drops straight into canyon (lots of bad outcomes)
- **Probabilistic planner** drops in flatter region gives up bestcase performance to protect against lots of poor outcomes

Example 3: Maneuver-Based Trajectory Planning

Use library of (uncertain) maneuvers to construct path that minimizes expected cost while satisfying chance constraints

 $\min_{\mu\in U} E[J(H_{\mu}(x_0,t_0))]$

s.t. $P(H_{\mu}(x_0, t_0) \notin F) \leq r$

 H_{μ} gives the state history under the controller μ

Koopman operator used to pull-back expected cost and constraint values for each maneuver

Gutow and Rogers, IEEE RAL, 2020.

Single Maneuver Under Parameter Uncertainty:

Each realization has probability of occurring given joint distribution on parameters or ICs

2.0

1.8

0.0

05

1.0

Downrange (m)

1.5

2.0

2.5

Example 3: Maneuver-Based Trajectory Planning

"Expected State Planner"

- Chain together next primitive from expected state of last one
- Use Koopman operator to pull back expected costs and constraint violations of candidate paths
- Use of primitives + Koopman allows UQ without real-time simulation

A* or dynamic programming can be used to solve for optimal path.

1 runs, 1.0% risk tolerance

Yields planner with tunable risk thresholds

Gutow and Rogers, IEEE RAL, 2020.

- Vehicle has 40% chance of being destroyed every 0.25 sec inside region 1
- Vehicle has 2.5% chance of being destroyed every 0.025 sec inside region 2
- Trajectory adapts based on risk tolerance

So far, we have tried to optimize vector of initial inputs given desired expected values of observables

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Problem statement:

Find
$$f_0(\mathbf{x})$$
 s.t.: $1 = \int_{supp(f_0)} f_0(\mathbf{x}) d\mathbf{x}$ Integrate to 1 constraint $c_i = \int_{supp(f_0)} f_0(\mathbf{x}) U_i g_i(\mathbf{x}) d\mathbf{x}$ $i = 1, \dots, p$ EV equality constraints $c_j < \int_{supp(f_0)} f_0(\mathbf{x}) U_j g_j(\mathbf{x}) d\mathbf{x}$ $j = p + 1, \dots, K$ EV inequality constraints

 U_i is Koopman operator that pulls observable function back from time t_i to t_0

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Meyers et al., J. Comp. Phys., 2021.

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This is an ill-posed problem. So we will need regularization.

Formulation as a quadratic program:

Approximate $f_0(x)$ as piecewise linear over grid

Quadrature approximation of desired EVs

Formulation as a quadratic program:

Non-negative constrained least-squares problem

Cast as a convex quadratic program

 $\operatorname{argmin} ||G\mathbf{f} - \mathbf{c}||_{2}^{2} + \lambda^{2} ||L\mathbf{f}||_{2}^{2}$ $\mathbf{f} \in \mathbb{R}^n$ Regularization EV targets (LS cost) $\mathbf{w}^T \mathbf{f} = 1$ Integrate to 1 constraint $G_{eq}\mathbf{f} = \mathbf{c}_{eq}$ $G_{ineq}\mathbf{f} \ge \mathbf{c}_{ineq}$ EV equality constraints EV inequality constraints $\mathbf{f} \ge \mathbf{0}$

Formulation as a quadratic program:

Non-negative constrained least-squares problem

Cast as a convex quadratic program

Use QP solver to find vector **f** which approximates initial distribution

Made possible because we formulated problem using Koopman expectations!

Vinh's Equations

| Case | Expected Value | Constraint |
|--------|---|---------------------|
| Case 1 | $Pr(1000 \le x(T) \le 1150 \text{ km})$ | ≥ 0.99 |
| | $Pr(0.09 \le V(T) \le 0.11 \text{ km/s})$ | ≥ 0.99 |
| | $Pr(Q(T) < 1.5 \times 10^7 \text{ kcal/m})$ | ≥ 0.99 |
| | $Pr(\max \dot{Q}(T) < 3 \times 10^5 \text{ kcal/m}^2/\text{s})$ | ≥ 0.99 |
| Case 2 | $Pr(1000 \le x(T) \le 1150 \text{ km})$ | ≥ 0.99 |
| | $Pr(0.09 \le V(T) \le 0.11 \text{ km/s})$ | ≥ 0.99 |
| | $E\left[\max \dot{Q}(T)\right]$ (kcal/m ² /s) | $= 3 \times 10^{5}$ |

Final **position** constraint Final **velocity** constraint Final **integrated heat load** constraint Maximum **heating rate** constraint (allowable range)

Maximum heating rate equality constraint

Uncertainty in lift coefficient (C_L), drag coefficient (C_D), heating coefficient (C_f)

What are allowable distributions for them?

Case 1: Allowable range of heating rates

Case 2: Maximum heating rate enforced

Meyers et al., J. Comp. Phys., 2021.

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Case 1: Allowable range of heating rates

Case 2: Maximum heating rate enforced

 Multi-dimensional distributions computed using 125,000 points (Case 1) and 15,625 points (Case 2)

0.01

Monte Carlo simulations verify that desired EV constraints were met using computed distributions

Conclusion

- Koopman operator provides powerful mechanism for optimization under parametric uncertainty
- Unique computational advantages compared to MC and other explicit UQ methods
- Approach has been demonstrated in optimization of discrete control decisions and initial uncertainty distributions
- Potential extensions to systems with process noise and cases involving optimization of continuous-time controllers

$$\int_{\Omega} \mathrm{P}_{\mathrm{S}} f(\boldsymbol{x}) g(\boldsymbol{x}) d\boldsymbol{x} = \int_{\Omega} f(\boldsymbol{x}) \mathrm{U}_{\mathrm{S}} g(\boldsymbol{x}) d\boldsymbol{x}$$

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